



## **NASA SBIR 2019 Phase I Solicitation**

### **Z4.02 In-Space Sub-Modular Assembly**

**Lead Center: LaRC**

**Participating Center(s): MSFC**

**Technology Area: TA12 Materials, Structures, Mechanical Systems and Manufacturing**

NASA envisions that persistent (very long duration) assets in space will require modular assembly architectures and interfaces to facilitate routine expansion, upgrade and refurbishment at the module and submodule level. This subtopic seeks novel approaches to three classes of module interface systems.

The first assembly need is autonomous (and highly automated) approaches and hardware concepts that support the interconnection of modules in the 100 – 5,000 kg range using some form of space robotics. (Note: The robotic manipulation systems are not the subject of this solicitation.) The objective of this first subtopic area is to minimize the parasitic mass from the joints and modularity features that are required for inter-module assembly. The lightweight connections between modules must include both electrical (power and data) and structural connections. Joining strategies that support fluid connections are of interest but not necessary to be responsive to this subtopic area. The structural connection should occur at a minimum of 3 discrete locations fixing the rigid body motion of the 2 modules in all 6 degrees of freedom while isolating (minimizing) forces resulting from thermal induced strain between the modules consistent with a LEO orbit. The three (or more) connections do not have to occur simultaneously.

The second assembly need is the development of a lightweight modular, palletizing system to support transport, emplacement and exchange of sub-modules in the 1-100 kg range. The palletizing system must support power and structural connections between the pallet and supporting backbone structure (fluid connections are a plus). Important considerations for the system are structurally efficient approaches that minimize parasitic mass, volume and power necessary to operate the palletizing system while minimizing forces resulting from temperature induced strain consistent with a LEO orbit.

The third assembly need is the development of assembly strategies that enable individual small spacecraft to support structural connects with suitable rigidity to form superstructures of small spacecraft (for the listed applications and other proposed science and exploration applications). Structural systems formed with assembled dimensions approximately 20x the size of the individual spacecraft are desired. Important consideration for the system are structurally efficient approaches that minimize parasitic mass, volume and power necessary to form assemblies utilizing approaches appropriate for small spacecraft, while minimizing forces resulting from temperature induced strain consistent with a LEO orbit.

NASA has plans to purchase services for delivery of payloads to the Moon through the Commercial Lunar Payload Services (CLPS) contract. Under this subtopic, proposals may include efforts to develop payloads for flight demonstration of relevant technologies in the lunar environment. The CLPS payload accommodations are yet to be precisely defined, however at least for early missions, proposed payloads should not exceed 15 kilograms in mass and not require more than 8 watts of continuous power. Smaller, simpler, and more self-sufficient payloads

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are more likely to be accommodated. Commercial payload delivery services may begin as early as 2020 and flight opportunities are expected to continue well into the future. In future years it is expected that payloads of higher mass and with higher power requirements might be accommodated. Selection for award under this solicitation will not guarantee selection for a lunar flight opportunity.

## Relevance to NASA

Fundamental to this work is the cross-cutting nature of the technologies to all NASA missions that benefit from multiple visits to an asset, both zero-g and planetary surface assets. The technologies developed under this topic have the potential to radically change the way missions and in-space capabilities are conceived and developed. Rapid emplacement of initial capability followed by systematic upgrade and expansion will increase investment value while simultaneously improving return on investment.

## References:

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